

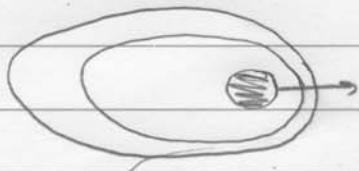
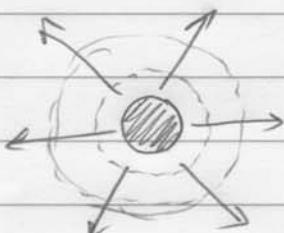
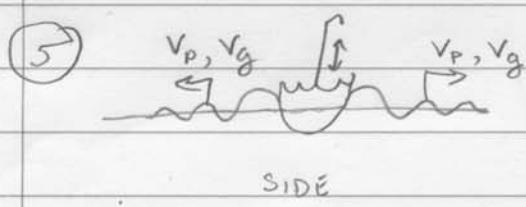
PART A -

①	X	0	0	0	0	X
	0	X	0	0	0	0
	0	0	X	X	0	0
	0	0	X	X	0	0
	0	0	0	0	0	0
	X	0	0	0	0	X

③	-X	0	0	0	0	-X
	0	-X	0	0	0	-X
	0	0	-X	+X	+X	0
	0	0	+X	-X	0	0
	0	0	+X	0	-X	0
	-X	-X	0	0	0	-X

② $\underline{\omega_s} = 16.75 \text{ rad/s}$, ($f_s = 2.67 \text{ Hz}$)

$$\omega_s = \omega_n = \sqrt{\frac{k}{m+ma}} \quad \text{lock in -}$$



$$V < V_g$$

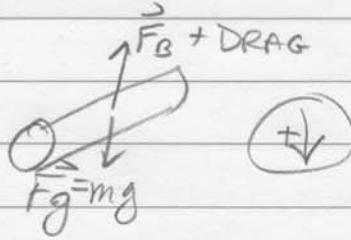


$$V > V_g$$

there is always some form of bow wave..

PART B-

(1) a) Terminal Velocity



$$F_{D\text{eng}} = \frac{1}{2} \rho u^2 C_D A$$

$$F_B = \rho g \cancel{A}$$

$$F_g = 1.5 \rho_w g \cancel{A}$$

$$A = \pi d L$$

$$\cancel{A} = \frac{\pi d^2}{4} L$$

$$@ \text{Terminal Vel } \sum F = ma = 0$$

$$\sum F = \vec{F}_g - \vec{F}_B - \vec{F}_D = 0$$

$$\vec{F}_D = \vec{F}_g - \vec{F}_B = \frac{1}{2} \rho_w g \cancel{A} = 157 N$$

$$u^2 \left(\frac{1}{2} \rho_w C_D d \cdot L \right) = \frac{1}{2} \rho_w g \frac{\pi d^2}{4} L$$

$$u = \sqrt{\frac{\pi g d}{4 C_D}} = \sqrt{157/C_D}$$

$$\text{laminar } C_D = 1.2 \quad u = 1.14 \text{ m/s} \quad Re = 228,000$$

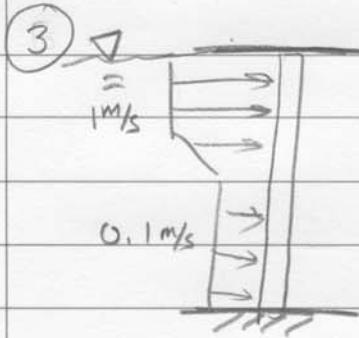
$$\text{turb } C_D = 0.6 \quad u = 1.62 \text{ m/s} \quad Re = 324,000$$

wow! v. close could be either so
it ultimately depends on cylinder
roughness - either answer gets credit-

$$b) @ u = 1.14 \text{ m/s} \quad f = 1.14 \text{ Hz} ; \quad f = \frac{s u}{d} \cdot \frac{0.2}{0.2} u$$

$$u = 1.62 \text{ m/s} \quad f = 1.62 \text{ Hz} ;$$

PART B



$$\text{assume } \delta = 0.2 = \frac{f_d}{u}$$

$$T_{\text{top}} \\ f = \frac{0.2 \cdot 1.0}{0.5}$$

$$\text{bottom} \\ f_{\text{bot}} = 0.1 * f_{\text{top}}$$

$$f_{\text{top}} = 0.4 \text{ Hz}$$

$$f_{\text{bot}} = 0.04 \text{ Hz}$$

b) Frequency of drag is 2. freq of vortex shedding
 freq. of lift is = freq of vortex shedding

Therefore it directly correlates w/ flow velocity.

c) The structure will tend to vibrate more at top exciting forced motion
 @ bottom. In both cases frequency is low so it's not going to be too violent - Shear will complicate shedding & make flow more 3 dimensional

d) Flexible cylinder

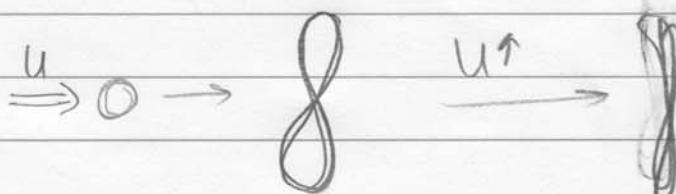


figure "8"
motion

higher vel, tension ↑
and inline vibe ↓

PART B-

(2) Initial Acceleration

$$\sum F = M \ddot{x}_3 |_{t=0} = \vec{F}_g - \vec{F}_s \quad \text{only forces at work!}$$

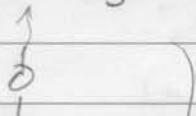
$$M = m + m_a \quad m = \rho V = 2\rho_{\text{air}} V$$

$V = V_{\text{sphere}} + 2V_{\text{cylinders}}$

$$= \frac{4}{3}\pi(2a)^3 + 2 \cdot \pi \left(\frac{a}{2}\right)^2 (4a)$$

$$= \frac{32}{3}\pi a^3 + 2\pi a^3 = \boxed{\frac{38}{3}\pi a^3 = V}$$

(i) vertical



$$m_a \underset{\substack{\text{only} \\ \downarrow}}{\approx} M a_{\text{sphere}} + \underset{\substack{\text{negligible} \\ \text{compared} \\ \text{to sphere} \\ \rightarrow}}{0} a_{\text{sphere}}$$

$$= \frac{1}{2} \rho_w V_s = \underline{\underline{\frac{16}{3} \rho_w \pi a^3}}$$

(ii) horizontal

$$m_a \underset{n}{=} \frac{1}{2} \rho V_s + 2 \pi \left(\frac{a}{2}\right)^2 (4a) \rho_w$$

$$M a_h \underset{\substack{\text{cylinders} \\ \rightarrow}}{=} \underline{\underline{\frac{22}{3} \pi \rho_w a^3}}$$

(↓↑)

$$(m+ma) \ddot{x}_3 = mg - \rho_w g V$$

$$= 2\rho_w g V - \rho_w g V$$

$$\ddot{x}_3 = \frac{\rho_w g V}{m+ma}$$

$$= \frac{\rho_w g \frac{38}{3}\pi a^3}{\frac{76}{3}\rho_w \pi a^3 + \frac{16}{3}\rho_s \pi a^3}$$

$$(M+ma) \ddot{x}_3 = mg - \rho_w g V = 1\rho_w g V$$

$$\ddot{x}_3 = \frac{\rho_w g \left(\frac{38}{3}\pi a^3\right)}{\left(\frac{76}{3} + \frac{22}{3}\right)\rho_w \pi a^3}$$

$$= \frac{38}{98} g$$

accel
is + down)

$$\ddot{x}_3 = \frac{38g}{92} = \frac{19}{46} g \text{ m/s}^2$$

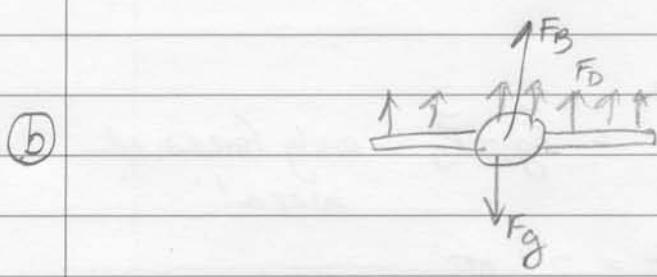
lower added mass
so slightly danger
initial acceleration

$$\ddot{x}_3 = \frac{19}{49} g \text{ m/s}^2 (\downarrow +)$$

Slightly lower due to
higher added mass

$$\ddot{x}_3 = 4.13 \text{ m/s}^2 (\downarrow +)$$

$$\ddot{x}_3 = 3.88 \text{ m/s}^2 (\downarrow +)$$



Terminal Vel $\Sigma F = 0 = m \ddot{x}_3 \Rightarrow \dot{x}_3 = 0$ no added mass per se.

$$mg - \rho g V - \frac{1}{2} \rho U^2 C_D A = 0$$

$$\rho g V = \frac{1}{2} \rho U^2 (C_D A) \quad U^2 = \frac{\rho g V}{\frac{1}{2} \rho C_D A} = \frac{2gV}{(C_D A)}$$

$$(C_D A) = A_{\text{sphere}} C_{D_{\text{sphere}}} + 2A_{\text{cyl}} C_{D_{\text{cyl}}}$$

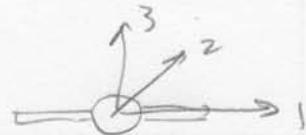
$$= 4\pi a^2 \cdot C_{D_s} + 2a(4a) C_{D_c}$$

$$U = \sqrt{\frac{2g \cdot \frac{38}{3} \pi a^2}{24a^2(\pi C_{D_s} + 2C_{D_c})}} = \sqrt{\frac{19g\pi a}{3(\pi C_{D_s} + 2C_{D_c})}}$$

if a small ($a \rightarrow 0$) $U \rightarrow 0$

$$\text{a large then } U = \sqrt{\frac{19g\pi a}{3(\pi C_{D_s} + 2C_{D_c})}}$$

$(C_D$, chosen based on $Re \#$)



$$\textcircled{c} \quad M_1 = ? - \epsilon_{ijk} u_i u_k m_j$$

$$u = (0, -u_2, u_3, 0, 0, 0)$$

$$M_2 = ?$$

no accelerations

$$\boxed{M_3 = 0} \text{ by symmetry}$$

on angular
velocities!

$$M_1 = -\epsilon_{123}^{(+1)} u_2 u_3 m_{33} - \epsilon_{132}^{(1)} u_3 u_2 m_{22} \quad \text{Munk moment only!}$$

$$j=1$$

$$(k=2,3) \quad \begin{matrix} 1 \\ 2 \\ i=2,3 \\ l=1-3 \end{matrix} \quad \begin{matrix} 2 \\ 3 \\ 3 \\ 3 \end{matrix}$$

$$\begin{matrix} 1 \\ 2 \\ 3 \\ 3 \\ k \end{matrix}$$

$$\begin{cases} m_{11}, m_{22}, m_{33} \neq 0 \\ m_{55}, m_{66} \neq 0 \\ \text{Rest are zero!} \end{cases}$$

$$m_{ji} = \begin{matrix} 1 & 2 & 1 & 3 \\ 2 & 2 & 2 & 3 \\ 3 & 2 & 3 & 3 \end{matrix}$$

$$M_1 = -1(u_2 u_3)[m_{22} - m_{33}] = 0$$

$$j=2 \Rightarrow$$

$$\boxed{M_2 = 0}$$

$$-\epsilon_{2kl}^j$$

$$j=2$$

bc $m_{22} = m_{33}$ by symmetry!

$$k=3 \text{ only}$$

$$l=1 \text{ then}$$

$$i=1 \text{ only}$$

$$\text{but } u_1 = 0 \text{ so}$$

for
 $j=3$ then

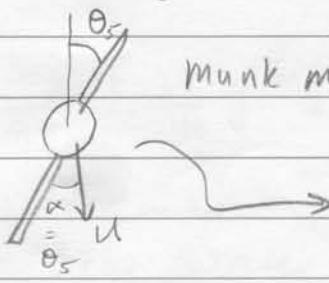
$$k=2 \text{ only}$$

$$l=1 \text{ only}$$

$$i=1 \text{ only}$$

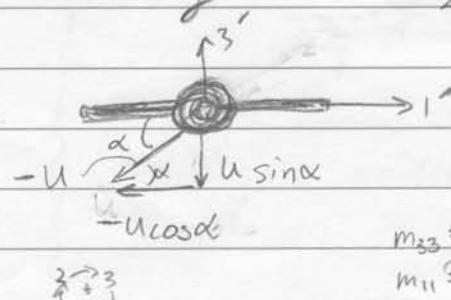
$$u_1 = 0 \text{ so } m_3 = 0$$

(d)



Munk moment only

$$u_x$$



$$u = (-u \cos \alpha, 0, u \sin \alpha, 0, 0, 0)$$

$$\begin{matrix} 2 \\ 3 \\ 1 \\ 2 \\ 3 \end{matrix}$$

$$m_{33} = \text{Mass from part a.}$$

$$m_{11} = \text{Mass from part a.}$$

$$\text{Want } M_2 = -\epsilon_{213} u_1 u_3 m_{33} - \epsilon_{231} u_3 u_1 m_{11} = u_1 u_3 [m_{33} - m_{11}] \neq 0$$

$$j=2 \quad k=1,3 \quad i=1,3 \quad l=1,3 \text{ for nonzero } m_{ijl}$$