

Goals for today

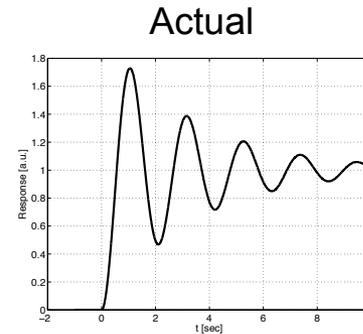
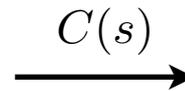
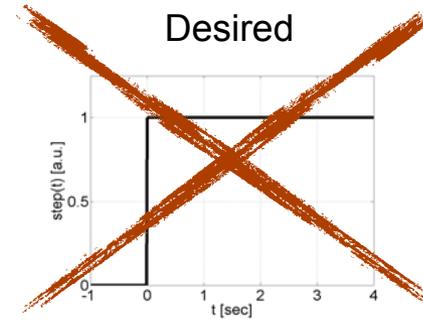
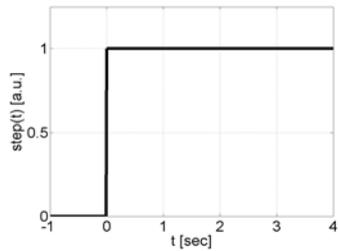
- Feedback topology
 - Negative vs positive feedback
- Example of a system with feedback
 - Derivation of the closed-loop transfer function
 - Specification of the transient response by selecting the feedback gain
- Steady-state error
 - General steady error determination
 - Steady-state error with external disturbance input
- MATLAB¹ LTI system analysis

Why feedback?

- Two reasons:
 - to make the system output resemble as much as possible a given input (“tracking”)
 - example: target-tracking missiles
 - to reduce the effect of disturbances in the system output
 - example: minute adjustments in steering wheel while we drive
- Today: we examine the first case only (tracking)
- We will discuss disturbances in a future lecture

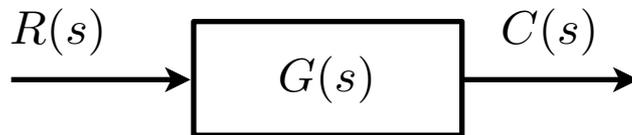
The problem with system dynamics

(for example)



Why?

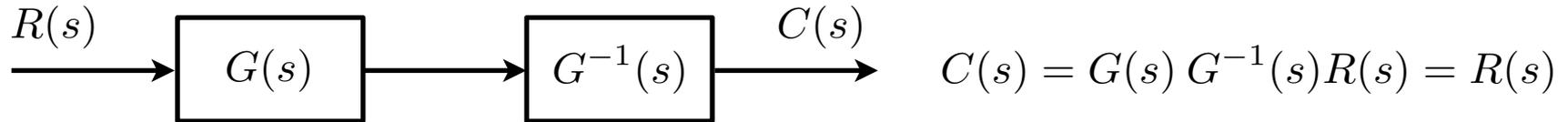
Because system response is determined by system transfer function $G(s)$



$$C(s) = G(s)R(s) \neq R(s)$$

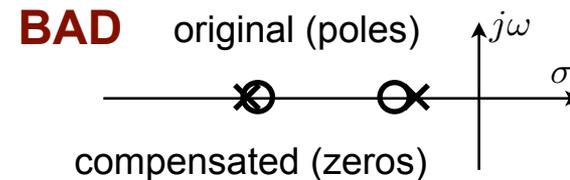
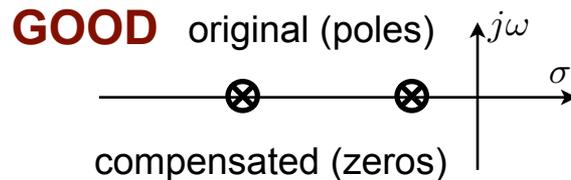
~~Suggested solution: invert the Transfer Function~~

Now indeed



Problems

- ➔ What if we don't know the transfer function exactly?
(or if there is some variation among systems manufactured in a factory)



- ➔ Systems with more zeros than poles are not physically realizable
(but they are realizable if we use digital implementation)

E.g. $G(s) = \frac{s}{(s + p_1)(s + p_2)}$

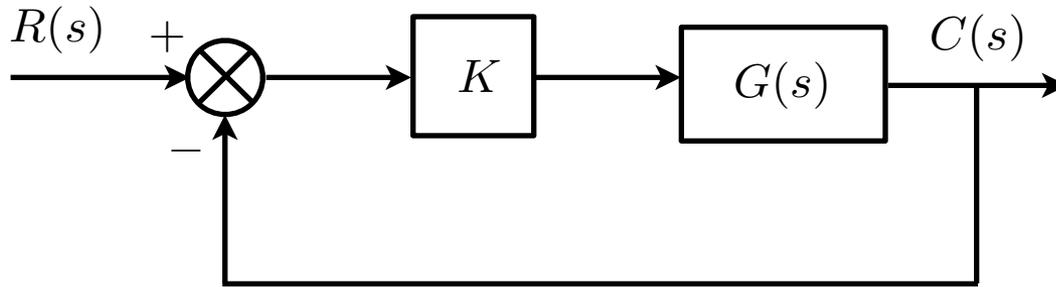
more poles

$$G^{-1}(s) = \frac{(s + p_1)(s + p_2)}{s}$$

more zeros

- ➔ Disturbances are amplified (we will talk about these later)

Instead: feedback (sort-of) “inverts” the TF



K : Feedback “gain”

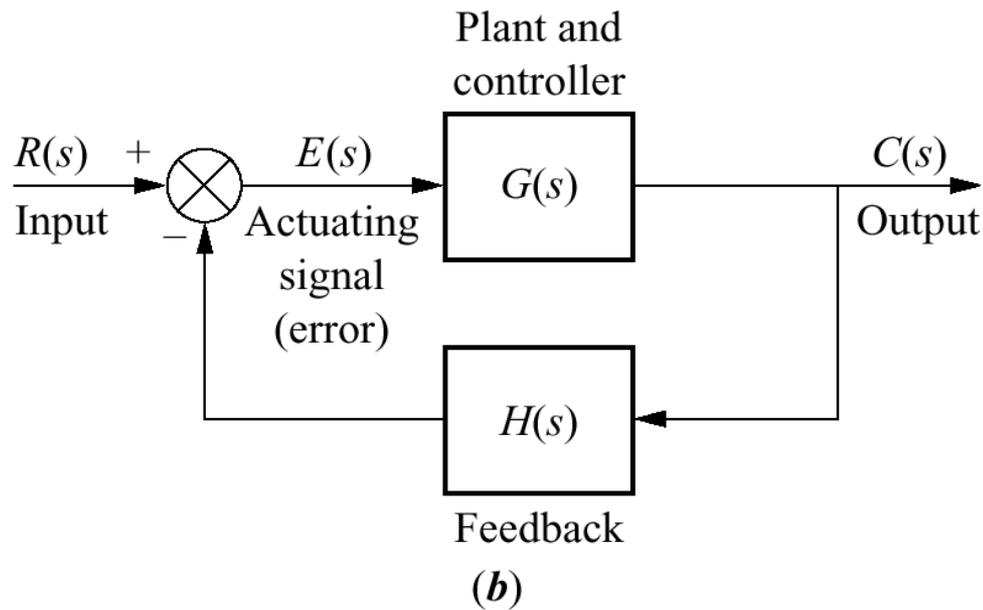
$G(s)$: “Open Loop (OL)” transfer function

“Closed Loop” transfer function $\frac{C(s)}{R(s)} = ?$

$$C(s) = KG(s) (R(s) - C(s)) \Rightarrow \frac{C(s)}{R(s)} = \frac{KG(s)}{1 + KG(s)}$$

Note: if $K \rightarrow \infty \Rightarrow C(s) \approx R(s) \Rightarrow$ Tracking!!

Negative feedback loop with feedback controller



$$E(s) = R(s) - H(s)C(s)$$

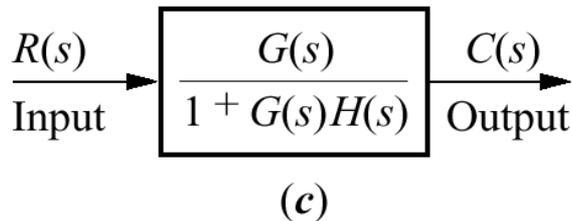
$$C(s) = E(s)G(s)$$

$$\Rightarrow C(s) = [R(s) - H(s)C(s)] G(s)$$

$$\Rightarrow \frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

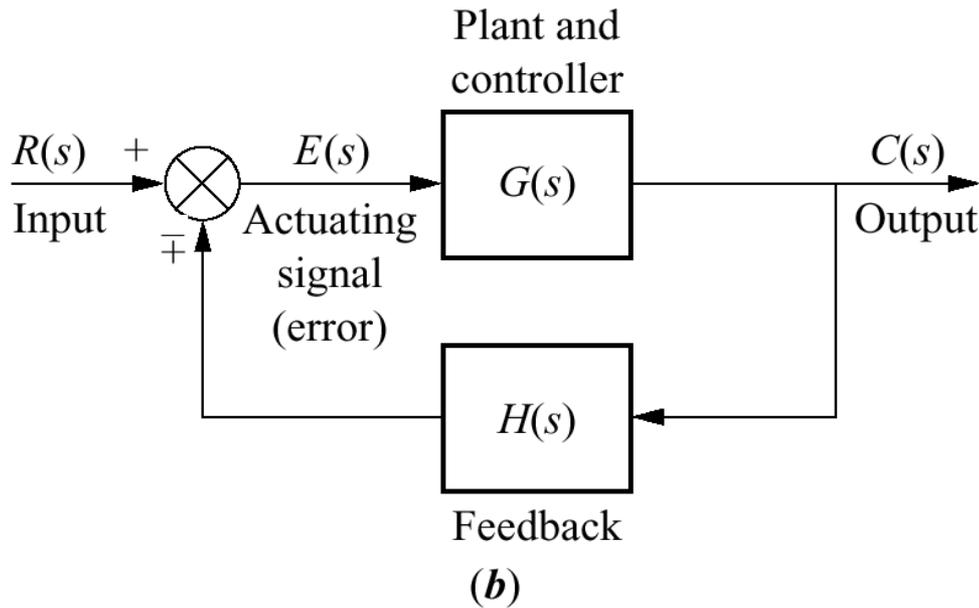
$\frac{C(s)}{R(s)}$: Closed-loop TF
 $G(s)H(s)$: Open-loop TF.

Equivalent system:



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Positive feedback



$$E(s) = R(s) + H(s)C(s)$$

$$C(s) = E(s)G(s)$$

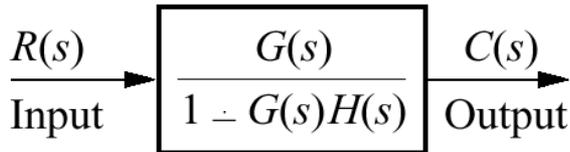
$$\Rightarrow C(s) = [R(s) + H(s)C(s)] G(s)$$

$$\Rightarrow \frac{C(s)}{R(s)} = \frac{G(s)}{1 - G(s)H(s)}$$

$$\frac{C(s)}{R(s)} : \text{Closed-loop TF}$$

$$G(s)H(s) : \text{Open-loop TF.}$$

Equivalent system:

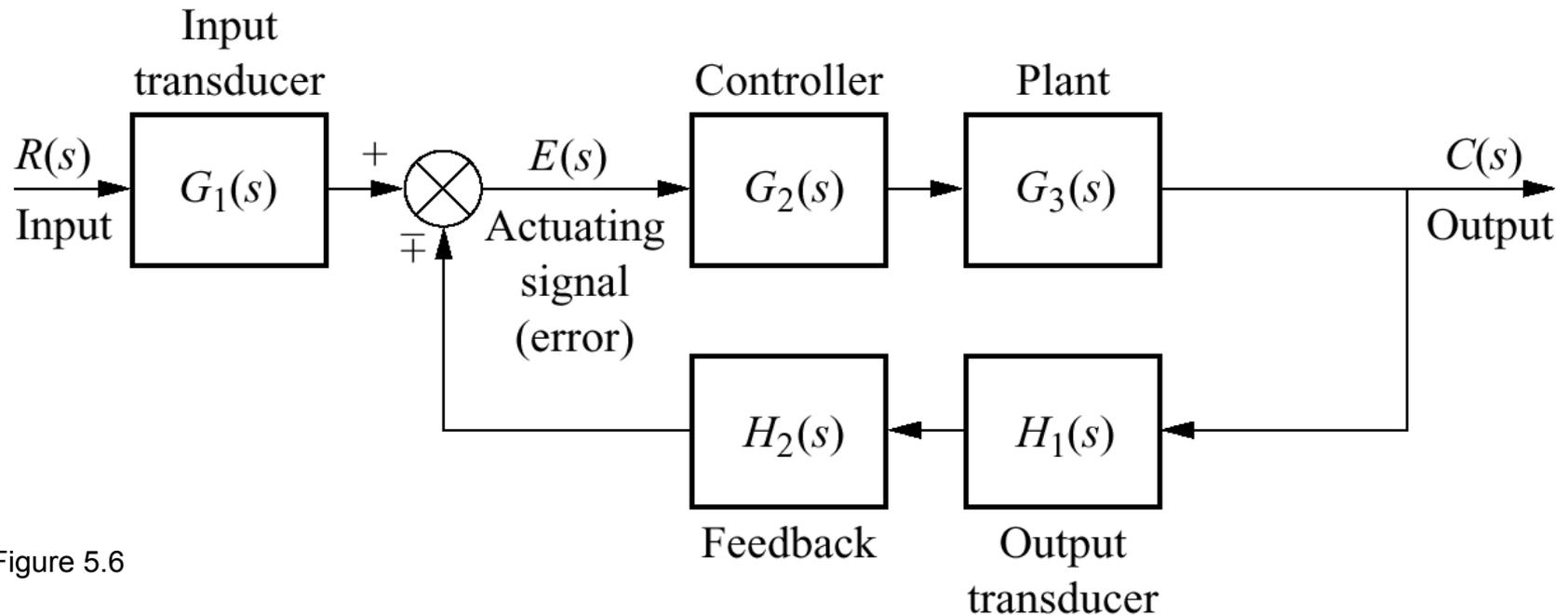


Nise Figure 5.6

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Generally, positive feedback is dangerous: it may lead to unstable response (i.e. exponentially increasing) if not used with care

A more general feedback system



Nise Figure 5.6

Plant: the system we want to control (e.g., elevator plant: input=voltage, output=elevator position)

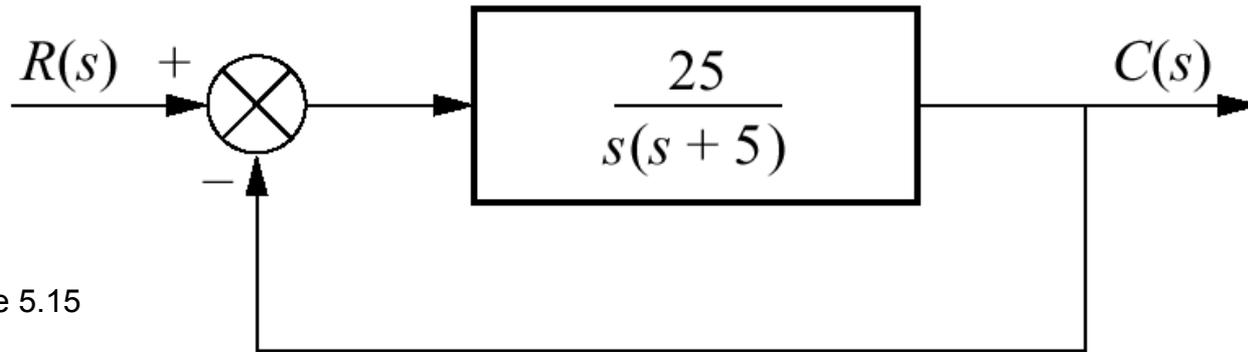
Controller: apparatus that produces input to plant (i.e. voltage to elevator's motor)

Transducers: converting physical quantities so the system can use them (e.g., input transducer: floor button pushed \rightarrow voltage; output transducer: current elevator position \rightarrow voltage)

Feedback: apparatus that contributes current system state to error signal (e.g., in elevator system, error=voltage representing desired position – voltage representing current position)

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Transient response of a feedback system

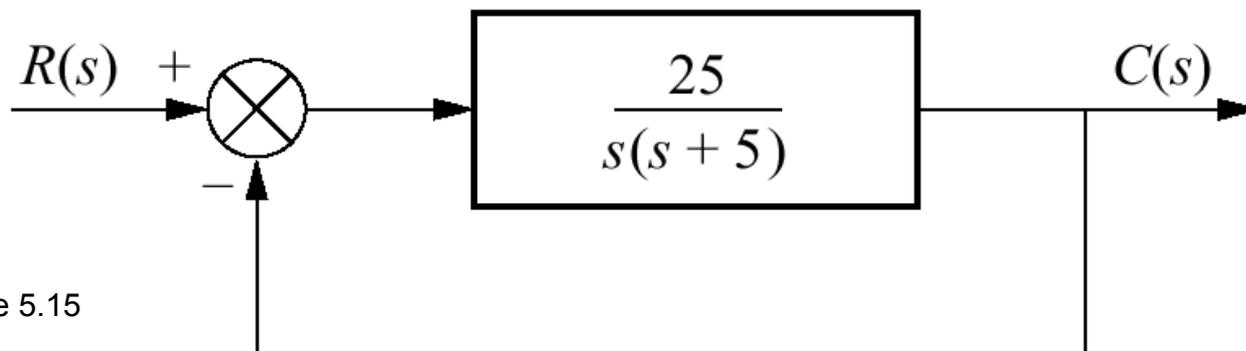


Nise Figure 5.15

Plant & Controller	$G(s)$	$= \frac{25}{s(s+5)}$,
Gain	K	$= 25,$
Feedback	$H(s)$	$= 1.$
Open loop TF	$G(s)H(s)$	$= \frac{25}{s(s+5)}$;
Closed loop TF	$\frac{G(s)}{1 + G(s)H(s)}$	$= \frac{25}{s^2 + 5s + 25}.$

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Transient response of a feedback system



Nise Figure 5.15

$$T(s) = \frac{G(s)}{1 + G(s)H(s)} = \frac{25}{s^2 + 5s + 25}$$

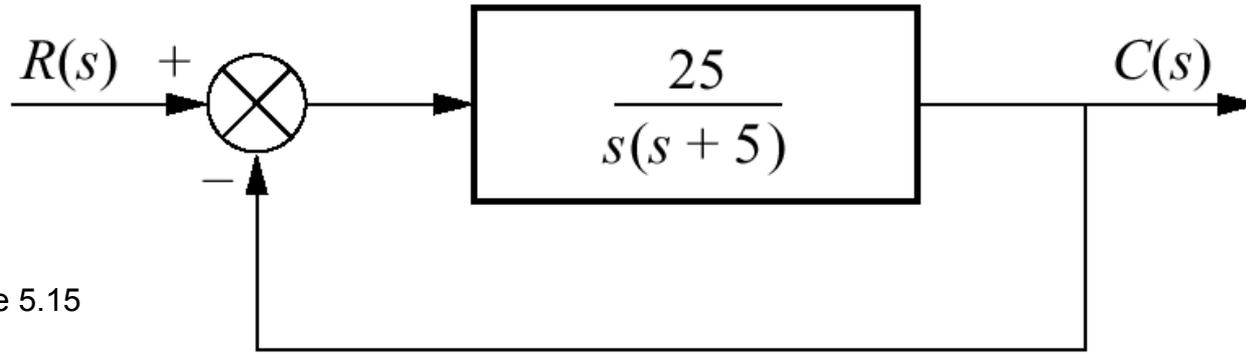
$$\text{Recall } \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$$\Rightarrow \omega_n = \sqrt{25} = 5 \text{ rad/sec}; \quad 2\zeta\omega_n = 5 \Rightarrow \zeta = 0.5;$$

\Rightarrow The feedback system is **underdamped**.

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Transient response of a feedback system



Nise Figure 5.15

$$T(s) = \frac{G(s)}{1 + G(s)H(s)} = \frac{25}{s^2 + 5s + 25}$$

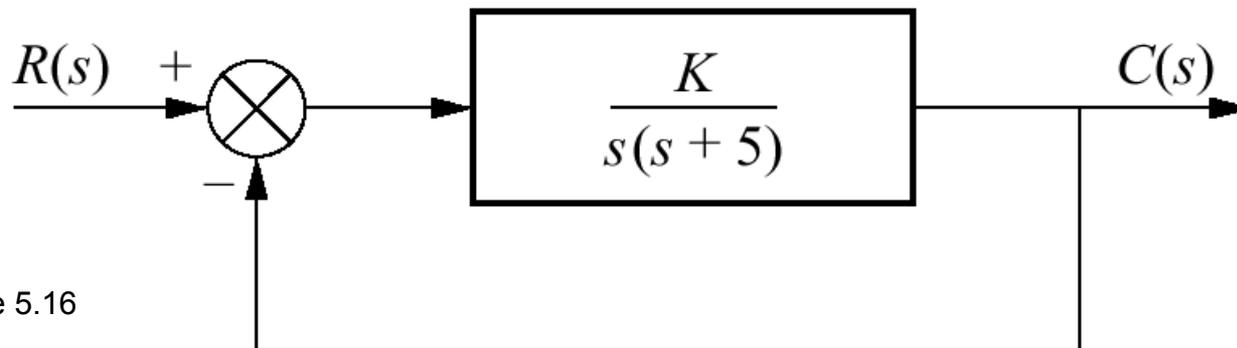
$$\text{Peak time } T_p = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}} = 0.726 \text{ sec,}$$

$$\%OS = \exp\left(-\frac{\zeta\pi}{\sqrt{1 - \zeta^2}}\right) \times 100 = 16.3,$$

$$\text{Settling time } T_s = \frac{4}{\zeta\omega_n} = 1.6 \text{ sec.}$$

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Adjusting the transient by feedback



Nise Figure 5.16

$$T(s) = \frac{G(s)}{1 + G(s)H(s)} = \frac{K}{s^2 + 5s + K}.$$

We wish to reduce overshoot to %OS=10% or less.

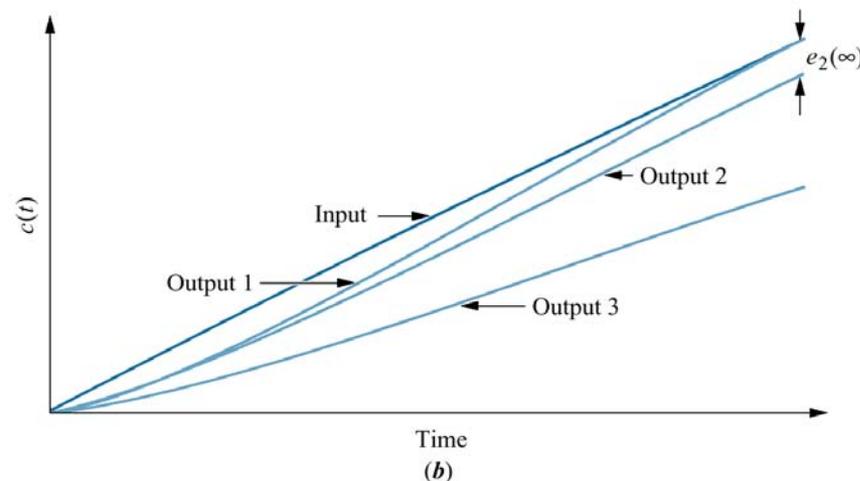
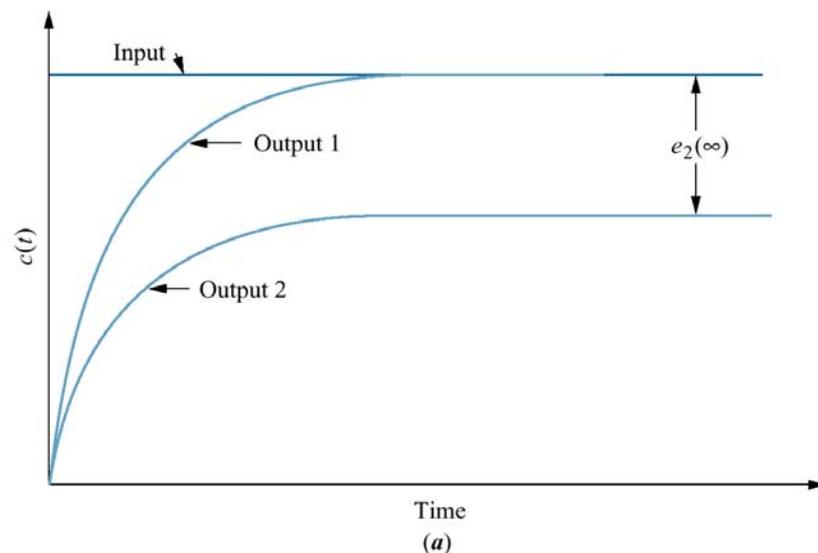
$$\omega_n = \sqrt{K}; \quad 2\zeta\omega_n = 5 \quad \Rightarrow \quad \zeta = \frac{5}{2\sqrt{K}}.$$

For 10% overshoot or less, we need $\zeta = 0.591$ or more. Therefore,

$$K = 17.9 \quad \text{or less.}$$

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Generalizing: steady-state error for arbitrary input



- Unit step input:
Steady-state error =
unit step – output as $t \rightarrow \infty$

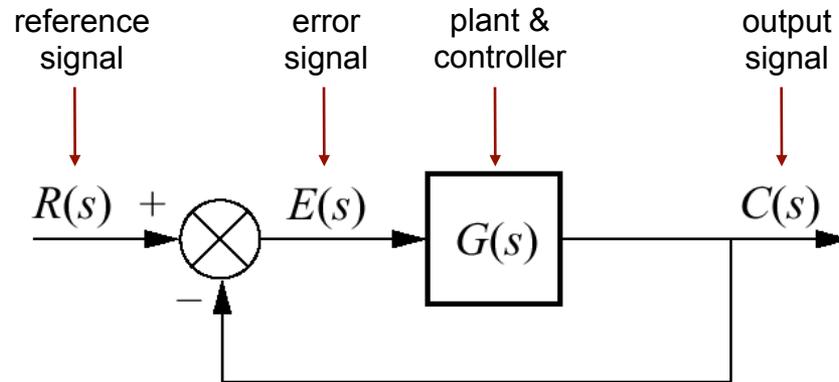
- Ramp input:
Steady-state error =
ramp – output as $t \rightarrow \infty$

Generally, the steady-state error is defined as

$$e(\infty) = \lim_{t \rightarrow \infty} [r(t) - c(t)] = \lim_{s \rightarrow 0} s [R(s) - C(s)],$$

where the last equality follows from the final value theorem.

Generalizing: steady-state error for arbitrary system, unity feedback



From the definition of the steady-state error,

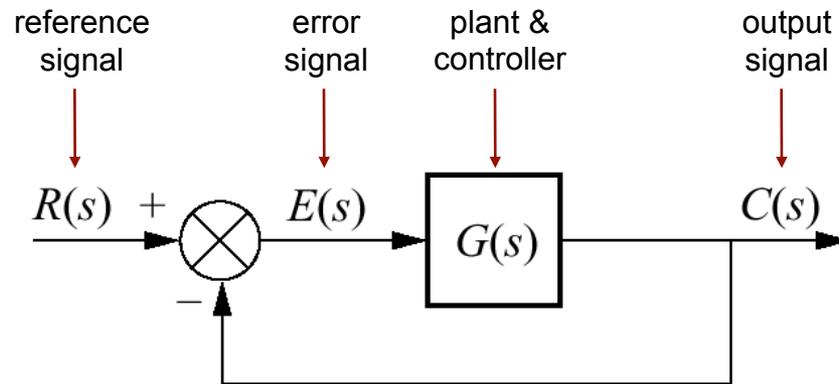
$$e(\infty) = \lim_{s \rightarrow 0} s \left[R(s) - C(s) \right] = \lim_{s \rightarrow 0} s E(s).$$

From the block diagram we can also see that

$$\frac{C(s)}{E(s)} = G(s) \Rightarrow E(s) = \frac{C(s)}{G(s)}.$$

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Generalizing: steady-state error for arbitrary system, unity feedback



Recall the closed-loop TF of the unity feedback system

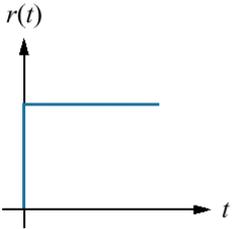
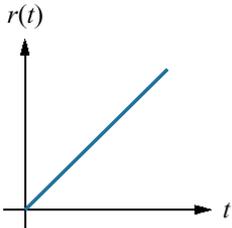
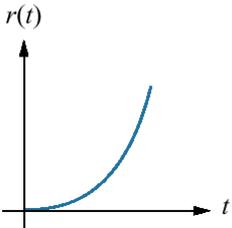
$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)} \Rightarrow C(s) = \frac{R(s)G(s)}{1 + G(s)}.$$

Substituting into the two formulae from the previous page,

$$E(s) = \frac{R(s)}{1 + G(s)} \Rightarrow e(\infty) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{sR(s)}{1 + G(s)}.$$

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Steady-state error and static error constants

Waveform	Name	Physical interpretation	Time function	Laplace transform
	Step	Constant position	1	$\frac{1}{s}$
	Ramp	Constant velocity	t	$\frac{1}{s^2}$
	Parabola	Constant acceleration	$\frac{1}{2}t^2$	$\frac{1}{s^3}$

$$e(\infty) = \lim_{s \rightarrow 0} \frac{sR(s)}{1 + G(s)}$$

$$\begin{aligned} e(\infty) &= \lim_{s \rightarrow 0} \frac{1}{s} \times \frac{s}{1+G(s)} \\ &= \lim_{s \rightarrow 0} \frac{1}{1+G(s)} \\ &= \frac{1}{1+\lim_{s \rightarrow 0} G(s)} \equiv \frac{1}{1+K_p} \end{aligned}$$

where $K_p = \lim_{s \rightarrow 0} G(s)$.

$$\begin{aligned} e(\infty) &= \lim_{s \rightarrow 0} \frac{1}{s^2} \times \frac{s}{1+G(s)} \\ &= \lim_{s \rightarrow 0} \frac{1}{s+sG(s)} \\ &= \frac{1}{\lim_{s \rightarrow 0} sG(s)} \equiv \frac{1}{1+K_v} \end{aligned}$$

where $K_v = \lim_{s \rightarrow 0} sG(s)$.

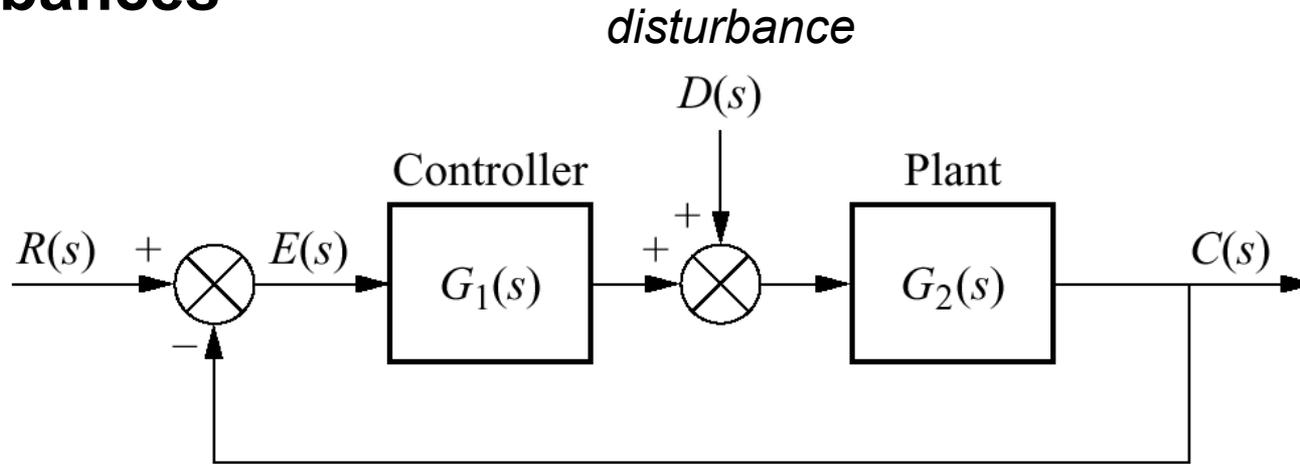
$$\begin{aligned} e(\infty) &= \lim_{s \rightarrow 0} \frac{1}{s^3} \times \frac{s}{1+G(s)} \\ &= \lim_{s \rightarrow 0} \frac{1}{s^2+s^2G(s)} \\ &= \frac{1}{\lim_{s \rightarrow 0} s^2G(s)} \equiv \frac{1}{1+K_a} \end{aligned}$$

where $K_a = \lim_{s \rightarrow 0} s^2G(s)$.

Note: the system must be **stable** (*i.e.*, all poles on left-hand side or at the origin) for these calculations to apply

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Disturbances



From the I–O relationship of the plant,

$$C(s) = E(s)G_1(s)G_2(s) + D(s)G_2(s).$$

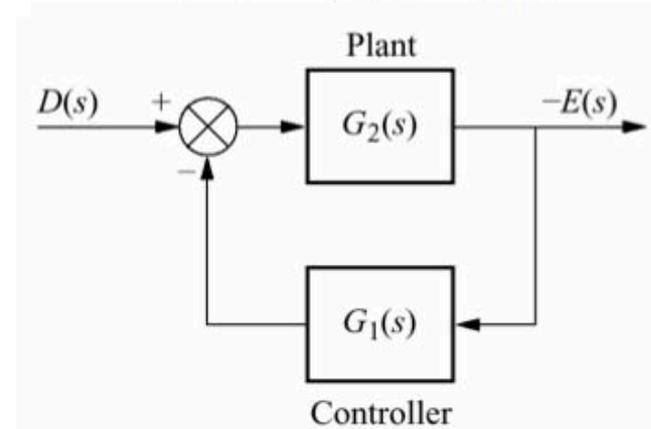
From the summation element,

$$E(s) = R(s) - C(s).$$

Substituting $C(s)$ and solving for $E(s)$,

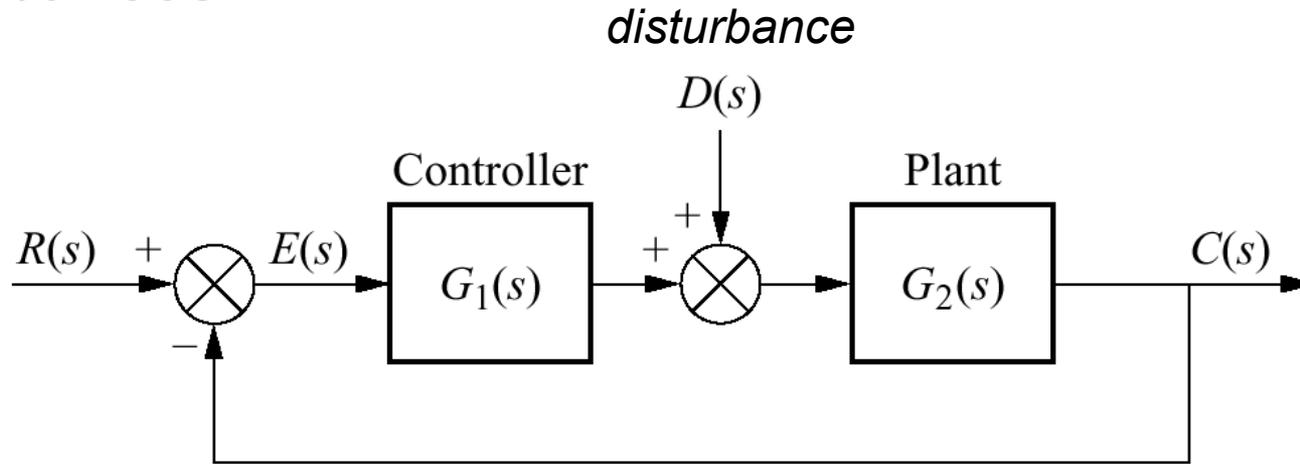
$$E(s) = R(s) \frac{1}{1 + G_1(s)G_2(s)} - D(s) \frac{G_2(s)}{1 + G_1(s)G_2(s)}$$

Equivalent block diagram
with $D(s)$ as input
and $-E(s)$ as output.



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Disturbances

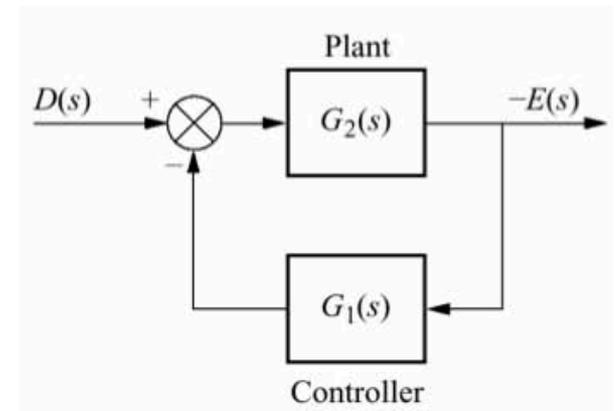


$$e(\infty) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} s \left[R(s) \frac{1}{1 + G_1(s)G_2(s)} - D(s) \frac{G_2(s)}{1 + G_1(s)G_2(s)} \right] \equiv e_R(\infty) + e_D(\infty),$$

where

$$e_R(\infty) = \lim_{s \rightarrow 0} \frac{sR(s)}{1 + G_1(s)G_2(s)}$$

$$e_D(\infty) = -\lim_{s \rightarrow 0} \frac{sG_2(s)D(s)}{1 + G_1(s)G_2(s)}$$



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2.04A Systems and Controls
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