

Topic 9

The Two-Noded Truss Element— Total Lagrangian Formulation

Contents:

- Derivation of total Lagrangian truss element displacement and strain-displacement matrices from continuum mechanics equations
- Mathematical and physical explanation that only one component (S_{11}) of the 2nd Piola-Kirchhoff stress tensor is nonzero
- Physical explanation of the matrices obtained directly by application of the principle of virtual work
- Discussion of initial displacement effect
- Comparison of updated and total Lagrangian formulations
- Example analysis: Collapse of a truss structure
- Example analysis: Large displacements of a cable

Textbook:

Section 6.3.1

Examples:

6.15, 6.16

TOTAL LAGRANGIAN FORMULATION OF TRUSS ELEMENT

We directly derive all required matrices in the stationary global coordinate system.

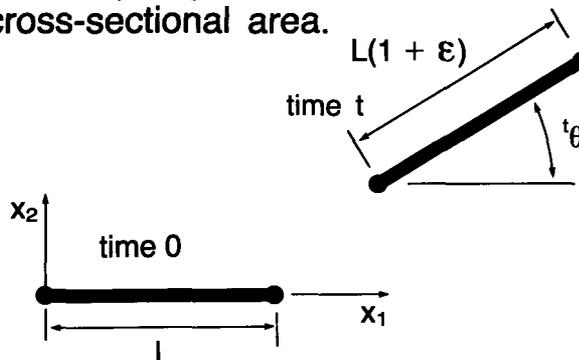
Recall that the linearized equation of the principle of virtual work is

$$\int_{0V} {}_0C_{ijrs} {}_0e_{rs} \delta {}_0e_{ij} {}^0dV + \int_{0V} {}^tS_{ij} \delta {}_0\eta_{ij} {}^0dV = {}^{t+\Delta t}\mathcal{R} - \int_{0V} {}^tS_{ij} \delta {}_0e_{ij} {}^0dV$$

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We will now show that the only non-zero stress component is ${}^tS_{11}$.

- 1) Mathematical explanation:
For simplicity, we assume constant cross-sectional area.



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We may show that for the fibers of the truss element

$${}^0\underline{X} = \begin{bmatrix} (1 + \epsilon) \cos^t\theta & -\sin^t\theta \\ (1 + \epsilon) \sin^t\theta & \cos^t\theta \end{bmatrix}$$

Since the truss carries only axial stresses,

$${}^t\underline{T} = \frac{{}^tP}{A} \begin{bmatrix} (\cos^t\theta)^2 & (\cos^t\theta)(\sin^t\theta) \\ (\cos^t\theta)(\sin^t\theta) & (\sin^t\theta)^2 \end{bmatrix}$$

written in the stationary coordinate frame

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Hence using

$${}^0\underline{S} = \frac{{}^0\rho}{{}^t\rho} {}^0\underline{X} {}^t\underline{T} {}^0\underline{X}^T$$

we find

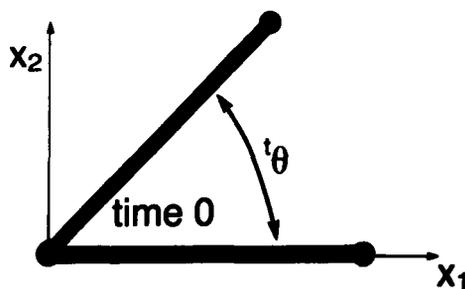
$${}^0\underline{S} = \begin{bmatrix} \frac{{}^tP}{A} \left(\frac{1}{1 + \epsilon} \right) & 0 \\ 0 & 0 \end{bmatrix}$$

1 for small ϵ

Physical explanation: we utilize an intermediate configuration t^*

time t^* (conceptual):
Element is stretched by tP .

time t : The element is moved as a rigid body.



$${}^0\underline{S} = {}^0\underline{T} = \begin{bmatrix} {}^0P/A & 0 \\ 0 & 0 \end{bmatrix}$$

$${}^{t^*}\underline{S} = {}^{t^*}\underline{T} = \begin{bmatrix} {}^{t^*}P/A & 0 \\ 0 & 0 \end{bmatrix}$$

$${}^t\underline{S} = \begin{bmatrix} {}^tP/A & 0 \\ 0 & 0 \end{bmatrix}$$

(the components of the 2nd Piola-Kirchhoff stress tensor do not change during a rigid body motion)

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The linearized equation of motion simplifies to

$$\int_{0V} {}^0C_{1111} {}^0e_{11} \delta_0 e_{11} {}^0dV + \int_{0V} {}^0S_{11} \delta_0 \eta_{11} {}^0dV$$

$$= {}^{t+\Delta t}\mathcal{R} - \int_{0V} {}^tS_{11} \delta_0 e_{11} {}^0dV$$

Again, we need only consider one component of the strain tensor.

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Next we recognize:

$${}^t\mathbf{S}_{11} = \frac{{}^tP}{A}$$

$${}^0C_{1111} = E, \quad {}^0V = A L$$

The stress and strain states are constant along the truss.

Hence the equation of motion becomes

$$\begin{aligned} (EA) {}^0\mathbf{e}_{11} \delta_0\mathbf{e}_{11} L + {}^tP \delta_0\eta_{11} L \\ = {}^{t+\Delta t}\mathcal{R} - {}^tP \delta_0\mathbf{e}_{11} L \end{aligned}$$

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To proceed, we must express the strain increments in terms of the displacement increments:

$${}^0\mathbf{e}_{11} = {}^0\mathbf{B}_L \underline{\hat{u}},$$

$$\delta_0\eta_{11} = (\delta \underline{\hat{u}}^T {}^t\mathbf{B}_{NL}^T) ({}^0\mathbf{B}_{NL} \underline{\hat{u}})$$

where

$$\underline{\hat{u}} = \begin{bmatrix} u_1^1 \\ u_2^1 \\ u_1^2 \\ u_2^2 \end{bmatrix}$$

Since ${}^0\epsilon_{11} = {}^0u_{1,1} + {}^0\overset{t}{u}_{1,1} {}^0u_{1,1} + {}^0\overset{t}{u}_{2,1} {}^0u_{2,1}$
 $+ \frac{1}{2} (({}^0u_{1,1})^2 + ({}^0u_{2,1})^2)$

we recognize

$${}^0e_{11} = {}^0u_{1,1} + {}^0\overset{t}{u}_{1,1} {}^0u_{1,1} + {}^0\overset{t}{u}_{2,1} {}^0u_{2,1}$$

$$\delta {}^0\eta_{11} = \delta {}^0u_{1,1} {}^0u_{1,1} + \delta {}^0u_{2,1} {}^0u_{2,1}$$

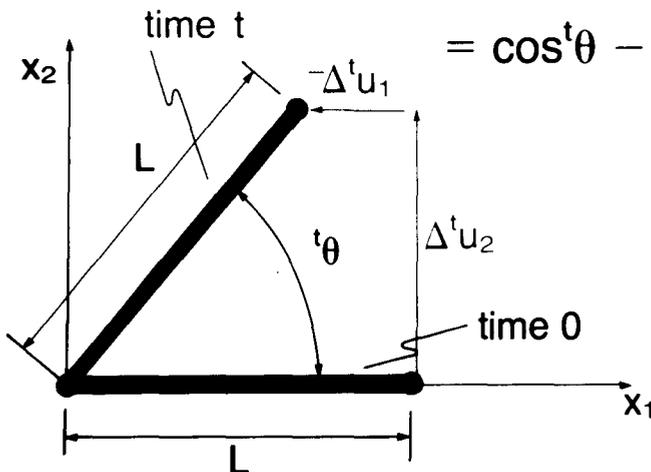
$$= [\delta {}^0u_{1,1} \quad \delta {}^0u_{2,1}] \begin{bmatrix} {}^0u_{1,1} \\ {}^0u_{2,1} \end{bmatrix}$$

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We notice the presence of ${}^0\overset{t}{u}_{1,1}$ and ${}^0\overset{t}{u}_{2,1}$ in ${}^0e_{11}$. These can be evaluated using kinematics:

$${}^0\overset{t}{u}_{1,1} = \frac{\Delta {}^t u_1}{L} \quad , \quad {}^0\overset{t}{u}_{2,1} = \frac{\Delta {}^t u_2}{L}$$

$$= \cos^t\theta - 1 \quad = \sin^t\theta$$



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We can now write the displacement derivatives in terms of the displacements (this is simple because all quantities are constant along the truss). For example,

$${}^0u_{1,1} = \frac{\partial u_1}{\partial {}^0x_1} = \frac{\Delta u_1}{\Delta {}^0x_1} = \frac{u_1^2 - u_1^1}{L}$$

Hence we obtain

$$\begin{bmatrix} {}^0u_{1,1} \\ {}^0u_{2,1} \end{bmatrix} = \frac{1}{L} \begin{bmatrix} -1 & 0 & 1 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \begin{bmatrix} u_1^1 \\ u_2^1 \\ u_1^2 \\ u_2^2 \end{bmatrix}$$

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Therefore

$$\begin{aligned} {}^0e_{11} &= {}^0u_{1,1} + [{}^t u_{1,1} \quad {}^t u_{2,1}] \begin{bmatrix} {}^0u_{1,1} \\ {}^0u_{2,1} \end{bmatrix} \\ &= \underbrace{\frac{1}{L} [-1 \quad 0 \quad 1 \quad 0]}_{{}^t B_{L0}} \hat{u} \\ &\quad + \underbrace{[\cos^t \theta - 1 \quad \sin^t \theta]}_{\text{initial displacement effect } {}^t B_{L1}} \left(\frac{1}{L} \begin{bmatrix} -1 & 0 & 1 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \right) \hat{u} \end{aligned}$$

$$\begin{aligned}
 {}_0e_{11} &= \frac{1}{L} \underbrace{\begin{bmatrix} -1 & 0 & 1 & 0 \end{bmatrix}}_{{}^t\mathbf{B}_{L0}} \underline{\hat{u}} \\
 &+ \frac{1}{L} \underbrace{\begin{bmatrix} -(\cos^t\theta - 1) & -\sin^t\theta & \cos^t\theta - 1 & \sin^t\theta \end{bmatrix}}_{{}^t\mathbf{B}_{L1}} \underline{\hat{u}} \\
 &= \frac{1}{L} \underbrace{\begin{bmatrix} -\cos^t\theta & -\sin^t\theta & \cos^t\theta & \sin^t\theta \end{bmatrix}}_{{}^t\mathbf{B}_L} \underline{\hat{u}}
 \end{aligned}$$

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Also

$$\delta_0\eta_{11} = \underbrace{\delta \underline{\hat{u}}^T}_{\begin{bmatrix} \delta_0 u_{1,1} & \delta_0 u_{2,1} \end{bmatrix}} \underbrace{\begin{pmatrix} \underbrace{{}^t\mathbf{B}_{NL}^T}_{\begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 1 & 0 \\ 0 & 1 \end{bmatrix}} \\ \frac{1}{L} \end{pmatrix}}_{{}^t\mathbf{B}_{NL}} \underbrace{\left(\frac{1}{L} \begin{bmatrix} -1 & 0 & 1 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \right)}_{\begin{bmatrix} {}_0u_{1,1} \\ {}_0u_{2,1} \end{bmatrix}} \underline{\hat{u}}$$

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Using these expressions,

$$(EA) \delta_0 e_{11} \delta_0 e_{11} L$$

$$\delta \underline{\hat{u}}^T \left(\frac{EA}{L} \begin{bmatrix} (\cos^4 \theta)^2 & (\cos^4 \theta)(\sin^4 \theta) & -(\cos^4 \theta)^2 & -(\cos^4 \theta)(\sin^4 \theta) \\ & (\sin^4 \theta)^2 & -(\cos^4 \theta)(\sin^4 \theta) & -(\sin^4 \theta)^2 \\ & & (\cos^4 \theta)^2 & (\cos^4 \theta)(\sin^4 \theta) \\ \text{symmetric} & & & (\sin^4 \theta)^2 \end{bmatrix} \right) \underline{\hat{u}}$$

$\underbrace{\hspace{15em}}_{{}^0K_L}$

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$${}^tP \delta_0 \eta_{11} L$$

$$\delta \underline{\hat{u}}^T \left(\frac{{}^tP}{L} \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 1 & 0 & -1 \\ -1 & 0 & 1 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix} \right) \underline{\hat{u}}$$

$\underbrace{\hspace{15em}}_{{}^0K_{NL}}$

and

$${}^t\mathbf{P} \delta_0 \mathbf{e}_{11} L$$

↓

$$\delta \underline{\hat{u}}^T \left({}^t\mathbf{P} \underbrace{\begin{bmatrix} -\cos^t\theta \\ -\sin^t\theta \\ \cos^t\theta \\ \sin^t\theta \end{bmatrix}}_{{}^t\mathbf{F}} \right)$$

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We notice that the element matrices corresponding to the T.L. and U.L. formulations are identical:

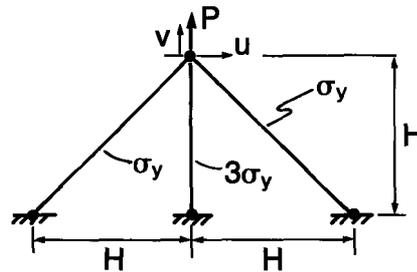
- The coordinate transformation used in the U.L. formulation is contained in the “initial displacement effect” matrix used in the T.L. formulation.
- The same can also be shown in detail analytically for a beam element, see K. J. Bathe and S. Bolourchi, Int. J. Num. Meth. in Eng., Vol. 14, pp. 961–986, 1979.

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Example: Collapse analysis of a truss structure

$$\begin{aligned} H &= 5 \\ A &= 1 \\ E &= 200,000 \\ E_T &= 0 \\ \sigma_y &= 100 \end{aligned}$$



- Perform collapse analysis using U.L. formulation.
- Test model response when using M.N.O. formulation.

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For this structure, we may analytically calculate the elastic limit load and the ultimate limit load. We assume for now that the deflections are infinitesimal.

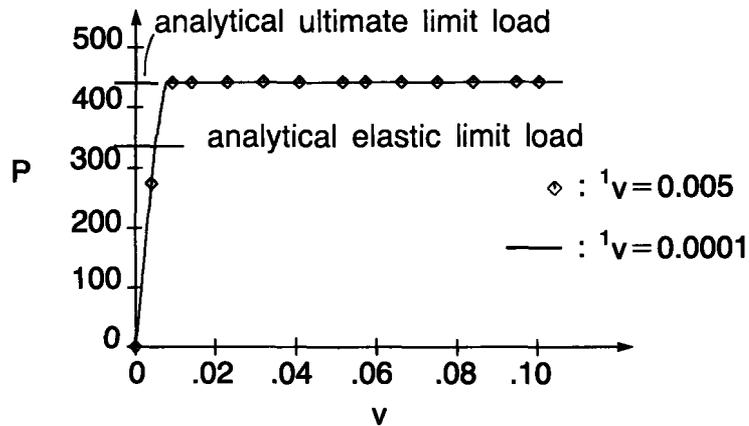
- Elastic limit load
(side trusses just become plastic)

$$P = 341.4$$

- Ultimate limit load
(center truss also becomes plastic)

$$P = 441.4$$

Using automatic load step incrementation and the U.L. formulation, we obtain the following results:



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We now consider an M.N.O. analysis.

We still use the automatic load step incrementation.

- If the stiffness matrix is not reformed, almost identical results are obtained (with reference to the U.L. results).

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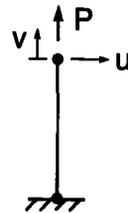
- If the stiffness matrix is reformed for a load level larger than the elastic limit load, the structure is found to be unstable (a zero pivot is found in the stiffness matrix).

Why?

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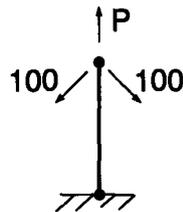
Explanation:

- In the M.N.O. analysis, once the side trusses have become plastic, they no longer contribute stiffness to the structure. Therefore the structure is unstable with respect to a rigid body rotation.



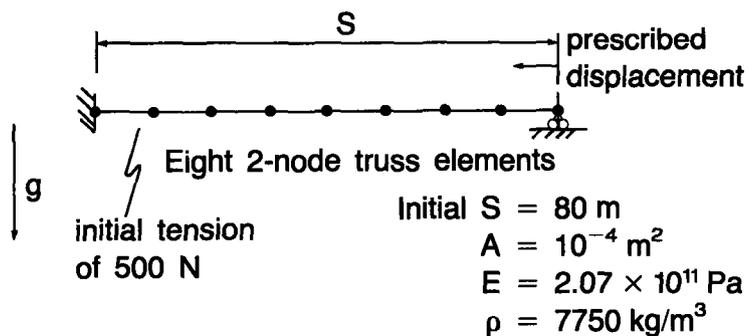
- In the U.L. analysis, once the side trusses have become plastic, they still contribute stiffness because they are transmitting forces (this effect is included in the \mathbf{K}_{NL} matrix).

Also, the internal force in the center truss provides stability through the \mathbf{K}_{NL} matrix.



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Example: Large displacements of a uniform cable



- Determine the deformed shape when $S = 30 \text{ m}$.

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This is a geometrically nonlinear problem (large displacements/large rotations but small strains).

The flexibility of the cable makes the analysis difficult.

- Small perturbations in the nodal coordinates lead to large changes in the out-of-balance loads.
- Use many load steps, with equilibrium iterations, so that the configuration of the cable is never far from an equilibrium configuration.

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Solution procedure employed to solve this problem:

- Full Newton iterations without line searches are employed.
- Convergence criteria:

$$\frac{\Delta \underline{U}^{(i)T} (\underline{R}^{t+\Delta t} - \underline{F}^{t+\Delta t(i-1)})}{\Delta \underline{U}^{(1)T} (\underline{R}^{t+\Delta t} - \underline{F}^t)} \leq 0.001$$

$$\|\underline{R}^{t+\Delta t} - \underline{F}^{t+\Delta t(i-1)}\|_2 \leq 0.01 \text{ N}$$

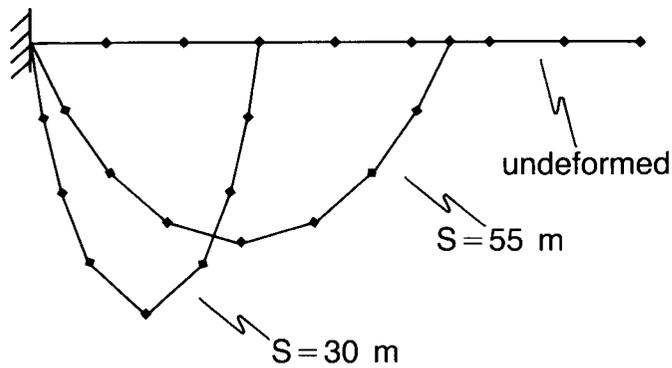
- The gravity loading and the prescribed displacement are applied as follows:

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Time step	Comment	Number of equilibrium iterations required per time step
1	The gravity loading is applied.	14
2-1001	The prescribed displacement is applied in 1000 equal steps.	≤ 5

Pictorially, the results are

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Klaus-Jürgen Bathe

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